

Question

Two-dimensional fluid flow takes place in the first quadrant of the (x, y) -plane. The stream function for the flow is given by

$$\psi(x, y) = Cxy$$

where C is a positive constant.

- (i) Determine the flow velocity components.
- (ii) Show that the flow is irrotational and incompressible.
- (iii) Sketch the streamlines of the flow

The stream function ψ is now regarded as the outer flow of a high Reynolds number steady viscous flow (with no body forces) and we wish to examine the boundary layer near the wall $y = 0$. Derive the (dimensional) boundary layer equations

$$\begin{aligned} uu_x + vu_y &= C^2x + vu_{yy} \\ u_x + v_y &= 0 \end{aligned}$$

where u and v are the velocity components of the flow, and give suitable boundary conditions for these equations.

Verify that a similarity solution exists in the form

$$\psi = xf(y)$$

and determine the differential equation satisfied by f , giving suitable boundary conditions.

Answer

We have $\phi = Cxy$, flow in $x \geq 0, y \geq 0$.

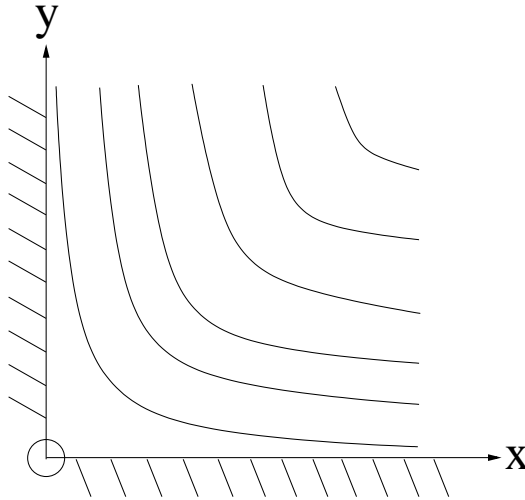
$$(i) \left. \begin{array}{l} u = \phi_y = Cx \\ v = -\phi_x = -Cy \end{array} \right\} \Rightarrow \underline{q} = (Cx, -Cy, 0)$$

$$(ii) \text{curl} \underline{q} = \begin{vmatrix} \frac{i}{\partial} & \frac{j}{\partial} & \frac{k}{\partial} \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ Cx & -Cy & 0 \end{vmatrix} = \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix} \Rightarrow \text{irrotational.}$$

$$\text{Also } \text{div}(\underline{q}) = \frac{\partial}{\partial x}(Cx) + \frac{\partial}{\partial y}(-Cy) + \frac{\partial}{\partial z}(0) = C - C = 0.$$

$$(iii) \phi = 0 \text{ on } x = 0, y = 0$$

$\phi = \text{constant} = \beta$ say $\Rightarrow xy = \text{constant} \Rightarrow$ hyperbolae.



Now consider the Navier-Stokes equations

$$\left. \begin{array}{l} \underline{q}_t + (\underline{q} \cdot \nabla) \underline{q} = \frac{-1}{\rho} \nabla p + \nu \nabla^2 \underline{q} \\ \text{div}(\underline{q}) = 0 \end{array} \right\} \text{Non - dimensionalise with } \underline{x} = L\bar{x}, \underline{q} = U\bar{q}, p = \rho U^2 \bar{p}$$

Where L and U are a representative length and speed.

(Dropping bars)

$$\frac{U^2}{L} (\underline{q} \cdot \nabla) \underline{q} = \frac{-U^2}{L} \nabla p + \frac{\nu U}{L^2} \nabla^2 \underline{q} = 0$$

$$\text{div}(\underline{q}) = 0$$

Thus the momentum equation becomes

$$(\underline{q} \cdot \nabla) \underline{q} = -\nabla p + \frac{1}{Re} \nabla^2 \underline{q}, \quad (Re = \frac{LU}{\nu})$$

Away from the boundaries in the flow, since $Re \gg 1$ the flow is essentially inviscid so that $\phi = Cxy$. But near $y = 0$ we must rescale $y = \delta \tilde{y}$, $v = \delta \tilde{v}$. ($\delta \ll 1$)

\Rightarrow

$$\begin{aligned} uu_x + \tilde{v}u_{\tilde{y}} &= -p_x + \frac{1}{Re} \left(u_{xx} + \frac{1}{\delta^2} u_{\tilde{y}\tilde{y}} \right) \\ \delta(u\tilde{v}_x + \tilde{v}\tilde{v}_{\tilde{y}}) &= \frac{1}{Re} \left(\delta\tilde{v}_{xx} + \frac{1}{\delta} \tilde{v}_{\tilde{y}\tilde{y}} \right) \\ u_x + \tilde{v}_{\tilde{y}} &= 0 \end{aligned}$$

Now consider the size of δ .

If $\delta^2 Re \ll 1$ then $u_{\tilde{y}\tilde{y}} = 0$ to leading order, and this can never match with the outer flow. If $\delta^2 Re \gg 1$ then we just get back to the inviscid equations.

$$\Rightarrow \delta^2 Re = 1, \text{ so } \delta = \frac{1}{\sqrt{Re}}.$$

Then the leading order equations (redimensionalized) are

$$\left. \begin{aligned} uu_x + vu + y &= \frac{-1}{\rho} p_x + \nu u_{yy} \\ 0 &= p_y \\ u_x + v_y &= 0 \end{aligned} \right\}$$

$$\text{Outer flow: } p + \frac{1}{2} p \underline{q}^2 = \text{constant}, \quad \Rightarrow p_x = -\rho u u_x$$

$$\text{But } u = Cx, \quad \Rightarrow -p_x = \rho C^2 x$$

$$\begin{aligned} \Rightarrow uu_x + vu_y &= C^2 x + \nu u_{yy} \\ u_x + v_y &= 0 \end{aligned}$$

Boundary conditions, $u = v = 0$ on $y = 0$ (no slip), $u \rightarrow Cx$ as $y \rightarrow \infty$ (matching).

$$\text{Now with } \phi = xf(y) \quad \begin{aligned} u &= xf' & u_y &= xf'' & u_{yy} &= xf''' \\ v &= -f & u_x &= f' \end{aligned}$$

$$\Rightarrow xf'^2 - fxf''' = C^2x + \nu xf'''$$

$$\Rightarrow f'^2 - ff''' - C^2 - \nu f''' = 0$$

$$\Rightarrow \nu f''' + ff''' - f'^2 + C^2 = 0$$

$$\text{Boundary conditions: } f(0) = f'(0) = 0, \quad f'(\infty) = c$$